

# A feasible multilayer structure design for solid lubricant coatings in a lunar environment†

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Solid lubricant coatings have received considerable research attention in space applications owing to their remarkably improved tribological characteristics. But their service life is seriously restricted by the harsh environment, such as high vacuum and abrasive wear. In this paper, a novel design of carbon-based multilayer (MoS<sub>2</sub>/DLC multilayer) coatings was reported to clarify the friction and wear behavior in high vacuum conditions with and without simulated lunar-dust (SLD). Compared with pure DLC or MoS<sub>2</sub> coatings, the multilayer coating showed excellent tribological performance with a low friction coefficient of 0.02 and a wear rate of  $\sim 6.5 \times 10^{-6} \text{ mm}^3 \text{ N}^{-1} \text{ m}^{-1}$ . What is particularly interesting is that the wear volume of MoS<sub>2</sub>/DLC multilayer coatings with the increase of time is in accordance with the Archard linear law, regardless of the condition with or without SLD. Moreover, the surface morphology and composition of wear tracks and scars reveal that the long life of carbon-based multilayer coatings cannot be explained solely by excellent mechanical performance, and is also attributed to the formation of ridge layers as third body reservoirs and a tribo-induced composite transfer layer containing SLD nanoparticles and coating materials.

## 1. Introduction

Many space-born systems involve relative motion of contacting surfaces, whose reliability is severely limited by the degradation of lubricants and excessive wear from lunar-dust. In NASA's reports, "dust is the number one environmental problem on the moon", the dust on the lunar surface posed a surprisingly broad array of difficulties in the moon probe program, such as vision obscuration, equipment clogging, radiator performance degradation, seal failure, and abrasive wear, whereas abrasive wear was mainly destruction of solid particles.<sup>1,2</sup> Consequently, a specific task has been set up under the exploration program to deal with the effects of lunar dust on surface wear.

The tribological properties of solid lubricants such as graphite and the transition metal dichalcogenides (TMD-sulfides, selenides or tellurides of tungsten, molybdenum and niobium) are of technological interest for reducing friction in circumstances where liquid lubricants are impractical, such as in space.<sup>3-9</sup> Unfortunately, the tribological performance and lifetime of these flexible coatings under high vacuum condition

with lunar-dust have greatly limited by the mechanical damage from the angular and sharp particles. Currently, diamond-like carbon (DLC) coatings are often referenced as potential candidates for space application due to their high load bearing capability, good adhesive strength, and excellent tribological properties.<sup>10-20</sup> In the pioneering work of Enke *et al.*, hydrogenated DLC coating sliding against steel could provide much lower friction coefficient of 0.005–0.02 in a vacuum or an inert atmosphere (dry N<sub>2</sub> or dry Ar).<sup>10</sup> Carpick *et al.* revealed the passivation mechanism of low friction and wear of ultra-nanocrystalline diamond and tetrahedral amorphous carbon thin coatings. They suggested that the surface hydrogen or hydroxyl passivation could reduce friction and wear in ambient air by preventing carbon–carbon bonding across the sliding interface.<sup>19,20</sup> Moreover, Qi *et al.* reported the effect of sand on dry sliding wear of a-C:H coating.<sup>12</sup> The friction coefficient under the sand-dust condition was relatively lower than that under dry-sliding condition, and the values of the wear rate under sand-dust condition were in the same order of magnitude with that under dry sliding condition without sand. However, in a high vacuum condition, the gradually reduced hydrogen content of the DLC coating during sliding cycle causes the deletion of C–H in the friction interface, thus leading to the adhesion and cold welding between the DLC coating and the counterpart.<sup>14-17</sup> Fontaine *et al.* have carried out significant work on diamond-like carbon-based coatings for space tribology. They conclude that a higher hydrogen content may be associated with a higher contribution of the weak van der Waals interactions, whereas a lower hydrogen content is characteristic

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of carbon-based coatings whose friction level is similar to graphite in UHV.<sup>21,22</sup>

Recent investigations on multilayer coatings composed of hard and ductile layers show an increased resistance to severe environmental conditions and an increase of durability in friction tests, which attributes to their moderate residential stress, good adherence and proper hardness to toughness ratio.<sup>23–26</sup> However, the tribological performance of the multilayer coating system with third-body abrasive particles in high vacuum has not received enough attention and extensive research. Based on our previous study,<sup>27</sup> we successfully fabricated a series of compositionally and structurally modulated MoS<sub>2</sub>/DLC multilayer coatings with different bilayer periods by unbalanced magnetron sputtering. A fundamental aspect of this work is to utilize the features of a multilayer design and to identify the relationships between the comprehensive performance and bilayer period. Moreover, we investigated the tribological mechanisms of the MoS<sub>2</sub>/DLC multilayer coating under high vacuum condition with simulated lunar-dust (SLD). The low friction coefficient and wear rate are found to depend on the formation of ridge layer in wear track and a composite transfer layer on steel ball. The present paper takes the first step of a wide program whose final objective is to apply the carbon-based multilayer coatings on space components.

## 2. Experimental details

### 2.1. Deposition and characterization of the MoS<sub>2</sub>/DLC multilayer coatings

A series of MoS<sub>2</sub>/DLC multilayer coatings with different bilayer periods were deposited on silicon wafers and stainless steel substrates using medium frequency magnetron sputtering. The MoS<sub>2</sub>/DLC multilayer coating coated Si wafer was used for physical characterization, such as SEM, EDX and AFM. And multilayer coating coated stainless steel was used for tribological test. A graphite target (purity 99.99%), one chromium target (purity 99.99%) and one molybdenum-disulfide target (purity 99.9%) were set in cylindrical vacuum chamber wall as the sputtering source. Prior to the coating deposition, the substrates were first cleaned ultrasonically in acetone and alcohol bath for 20 min, respectively, and then dried in nitrogen followed by plasma etching with Ar<sup>+</sup> ions in a vacuum chamber to remove the surface contaminations. Subsequently, the Cr interlayer of approximately 0.2 μm in thickness was deposited on the substrate to improve the adhesion between the multilayer coatings and substrates. Then the MoS<sub>2</sub>/DLC multilayer coatings were prepared by alternative deposition of DLC layer (graphite target current: 3.5 A, chromium target current: 0.38 A) and MoS<sub>2</sub> layer (MoS<sub>2</sub> target current: 0.8 A) for bilayer deposition times of 15.8 h, 7.6 h, 5.2 h, 3.9 h, and 0.6 h, respectively. The as-deposited coatings with bilayer thicknesses of 6000 nm, 3100 nm, 2000 nm, 1450 nm, and 210 nm are defined as MoS<sub>2</sub>/DLC-2L, 4L, 6L, 8L and 40L, respectively. And the corresponding numbers of periods are 1, 2, 3, 4, and 20, respectively. Fig. 1 shows the cross-sectional SEM images and the corresponding AFM surface morphologies of as-deposited MoS<sub>2</sub>/DLC multilayer coatings. As shown in the cross-sectional images, the

MoS<sub>2</sub>/DLC multilayer coatings have a homogeneous and dense microstructure, which is related to the columnar structure of MoS<sub>2</sub> sublayer. The thicknesses of the multilayer coatings are determined to be approximately 6.0 μm except the MoS<sub>2</sub>/DLC-40L coating whose thickness is 4.2 μm. The difference in the thickness is attributed to the discrepancy of deposition rates. The SEM and AFM surface morphologies show that the surfaces of the MoS<sub>2</sub>/DLC multilayer coatings are uniform and smooth. As shown in the Fig. 1f, the coatings are composed of small spheres whose sizes decreased first and increased afterwards with the increase in bilayer period. The mechanical properties of the MoS<sub>2</sub>/DLC multilayer coatings, such as hardness and elastic modulus are summarized in Table 1. As shown in Table 1, the hardness and elastic modulus of multilayer coatings increased first and decreased afterwards with the increase in bilayer period. However, when the bilayer period increased to 20 (MoS<sub>2</sub>/DLC-40L coating), the value of hardness and elastic modulus reached to the maximum. The reason for the sharply increasing *H* and *E* is the development of a large interfacial dislocation density during the elasto-plastic transition to fully plastic yield. Moreover, the *H/E* ratio of MoS<sub>2</sub>/DLC-40L is lowest, which indicates the 40L coating with a low elastic recovery is vulnerable to dent damage from third-body abrasive particles. Fig. S1 in ESI† shows the evolution of friction coefficient as a function of sliding time against steel ball without SLD condition. The friction coefficient of the MoS<sub>2</sub>/DLC multilayer coatings decreased first and increased afterward with bilayer period. And the MoS<sub>2</sub>/DLC-4L coating obtained the lowest value of 0.02. When the bilayer period increased to 20, the MoS<sub>2</sub>/DLC-40L coating only maintains a short lifetime (sliding about 18 min). This phenomenon attributes to the low *H/E* ratio of 40L coating (Table 1). By comparing the mechanical performance and the tribological behaviors of the MoS<sub>2</sub>/DLC multilayer coatings under vacuum condition indicate that the optimum bilayer period is 2 (MoS<sub>2</sub>/DLC-4L coating). Consequently, we will focus on the tribological behavior and wear mechanism of the MoS<sub>2</sub>/DLC-4L coating under high vacuum conditions with and without SLD.

### 2.2. Friction and wear test

The friction and wear properties of the MoS<sub>2</sub>/DLC multilayer coatings were examined by dry sliding under vacuum conditions with and without SLD. The artificial synthetic lunar-dust used as the abrasive particles were fabricated by the Institute of Aerospace 501 (Beijing, China). The tests were performed on a ball-on-disk configuration, where a steel ball of 4 mm diameter with hardness of 710 HV was loaded and rubbed against a multilayer coating. The loads of 3, 5, 7, and 10 N were applied through a stationary loading system, and the corresponding initial maximum Hertzian contact pressure were about 0.95, 1.12, 1.25, and 1.42 GPa. The rotate radius was 6 mm at 180 rotations per minute (rpm), which provided a velocity of 0.113 m s<sup>-1</sup>. Prior to the friction tests, a cylinder (∅ 10 cm × 10 cm) cans as the lunar-dust reservoir was placed in the vacuum chamber. The specimen was placed at the bottom of cans. The SLD particles were fully homogenized on the surface of selected lubricant coating, and

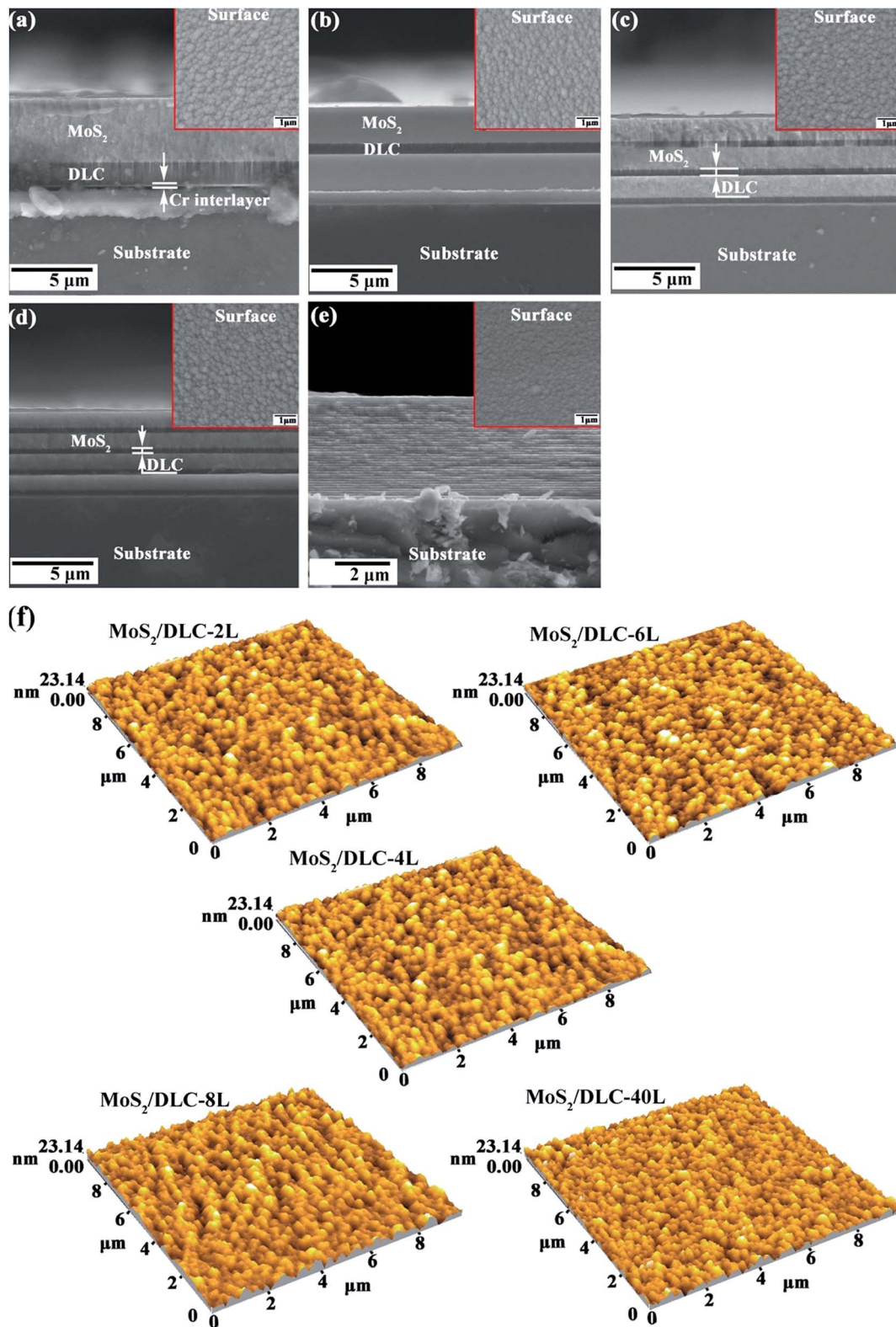


Fig. 1 Cross-sectional SEM images and corresponding surface topographies of as-deposited (a) MoS<sub>2</sub>/DLC-2L, (b) 4L, (c) 6L, (d) 8L, and (e) 40L coatings; (f) AFM surface morphologies of as-deposited multilayer coatings.

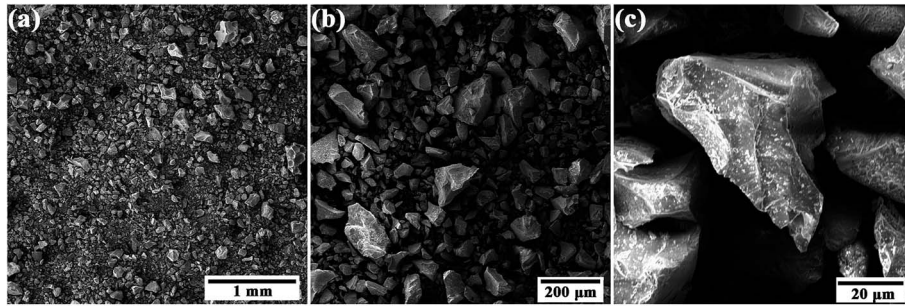
the counterpart steel ball preloaded on the SLD and the specimen holder remained still prior to the test. The specific amount of SLD particles are  $26.7 \pm 0.5 \text{ mg cm}^{-2}$ . Subsequently, the lid of cans

with some venthole and asbestos wire gauze was screwed on the top of cans. In the process of friction tests, the specimen was held stationary and the steel ball holder was rotating with velocity



**Table 1** Composition and mechanical properties of the MoS<sub>2</sub>/DLC multilayer coatings

Coating	Thickness (μm)	Roughness (nm)	Composition (at%)	Hardness and modulus (GPa)	<i>H/E</i>
2L	6.0	$R_a = 3.36, R_q = 4.20$	41.29 at% Mo, 57.71 at% S	<i>H</i> : 6.9, <i>E</i> : 82.4	0.084
4L	6.2	$R_a = 2.70, R_q = 3.38$	41.16 at% Mo, 58.84 at% S	<i>H</i> : 8.5, <i>E</i> : 107.6	0.079
6L	5.9	$R_a = 3.08, R_q = 3.83$	41.00 at% Mo, 59.00 at% S	<i>H</i> : 6.8, <i>E</i> : 78.8	0.086
8L	5.8	$R_a = 3.18, R_q = 3.98$	32.60 at% Mo, 67.40 at% S	<i>H</i> : 6.2, <i>E</i> : 81.8	0.075
40L	4.2	$R_a = 3.08, R_q = 3.86$	12.80 at% Mo, 21.18 at% S, 51.31 at% C, 14.71 at% Cr	<i>H</i> : 10.3, <i>E</i> : 150.7	0.068

**Fig. 2** SEM photomicrographs of the SLD particles. (a) 56 magnification, (b) 188 magnification, (c) 2000 magnification.

relative to the stationary contact area. The friction coefficient is continuously recorded during testing. The microphotographs of SLD with different magnification in Fig. 2 show the typical irregular shapes, including ‘uniquely shaped’, ‘fine grain angular fragments’, and ‘rodlike with bulbous ends’. The chemical compositions of SLD are shown in Table 2. In general, the shapes and compositions of SLD particles are similar to the real lunar-dust. All the friction tests were performed at room temperature and the vacuum chamber pressure less than  $1.0 \times 10^{-3}$  Pa. After friction test, the surface profile of wear tracks was measured using a surface profiler (D-100, KLA, Tencor). The wear volume of each wear track was derived from three to five measurements to calculate the average value. The wear rates of the coatings were calculated from their wear volumes using the following equation:

$$K = V/SF \quad (1)$$

where  $V$  is the wear volume in stere,  $S$  is the total sliding distance in meter, and  $F$  is the normal load in Newtons.

### 2.3. General characterization

The nanohardness ( $H$ ) and Young’s modulus ( $E$ ) of the coatings were measured using a Nanoindenter II microprobe with a diamond Berkovich (three-sided pyramid) indenter tip and the indentation depth was about 10% of the coating thickness to

minimize the substrate contribution. The internal stress of the coatings was checked by measuring the substrate bending using a profilometer and the Stoney’s equation. Wear tracks of the coatings and counterpart steel balls were examined using JEM-5600LV scanning electron microscopes (SEM; JEOL, Japan), energy dispersive X-ray analysis (EDX), Raman spectroscopy, and X-ray photoelectron spectroscopy (XPS). After the friction and wear measurements, the surface chemical composition were determined using scanning electron microscopy-energy dispersive X-ray spectroscopy (SEM-EDS) (Oxford IE250 Energy Dispersive Spectrometer, EDS) under 20 kV accelerating voltage with 10 nA beam current. The chemical states of the elements were analyzed by a PHI-5702 multifunctional X-ray photoelectron spectroscopy (XPS) made by American Institute of Physics Electronics Company using K-Alpha irradiation as the excitation source. The binding energies of the target elements were determined at a pass energy of 29.3 eV, with a resolution of about  $\pm 0.3$  eV, using the binding energy of contaminated carbon (C 1s: 284.8 eV) as the reference. The structural characteristics of the MoS<sub>2</sub>/DLC multilayer coatings before and after friction test were measured using Raman spectroscopy equipped with a 532 nm argon ion laser from 100 to 2000 cm<sup>-1</sup>. The wear debris was observed by FEI Tecnai F300 high-resolution transmission electron microscope (HRTEM) with an accelerating voltage of 300 kV.

## 3. Results and discussions

### 3.1. Effect of load on the friction behavior of MoS<sub>2</sub>/DLC multilayer coating

Fig. 3 shows the evolution of friction coefficient as a function of sliding time for the MoS<sub>2</sub>/DLC-4L coating against steel ball

**Table 2** Chemical composition of the simulant lunar-dust

Elements	O	Si	Al	Fe	Ca	Mg	Na	K	Ti
Composition (at%)	71.17	12.06	5.52	3.00	2.27	2.09	2.64	0.79	0.45

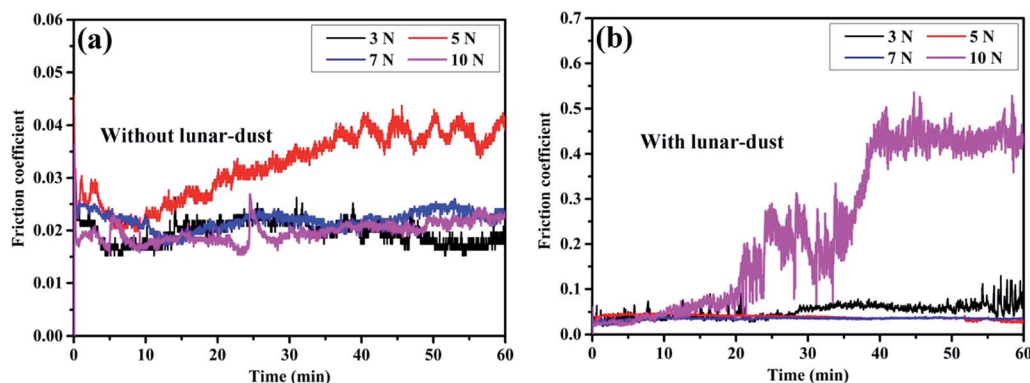


Fig. 3 Variation in friction coefficient of the MoS<sub>2</sub>/DLC-4L coating sliding against steel ball as a function of the sliding time under high vacuum conditions without (a) and with (b) SLD.

under vacuum conditions with and without SLD; and for the lubricated conditions without SLD (Fig. 3a), the variation trend in the friction coefficient over applied loads is not pronounced. However, for the condition with SLD, the friction coefficient fluctuated under 3 N and 10 N whereas maintained a steady value under 5 N and 7 N. Under a light load of 3 N, some of the SLD particles on the contact surface are not broken down and mainly exist in the contact area by a cutting mechanism; the variation and fluctuation of friction spikes for the MoS<sub>2</sub>/DLC-4L coating are associated with periodic loss of transfer layer due to the mechanical damage from sharp particles. With the increase in the applied load, the SLD particles could be crushed down and formed amounts of particles at small size to roll freely on the contact region. Moreover, the extremely small particles combining with coating material transferred to the steel ball to form a transfer layer, which was effective to reduce the friction coefficient. As a result, the friction coefficient stabilized gradually at a constant value. When the load was up to the maximum value of 10 N, the friction coefficient fluctuated remarkably ranging from 0.02 to 0.50. This behavior is due to the fact that

some abrasive particles of milled SLD embedded into the counterpart surface under heavy load and thus the transfer layer is difficult to form. Finally, the selected coating was completely removed after the periodic deformation and fatigue of 40 min.

Fig. 4 shows the SEM images of wear tracks for the MoS<sub>2</sub>/DLC-4L coating and the counterpart steel balls under different applied loads in a high vacuum environment without SLD. The sizes of wear tracks increase with in the applied loads. As shown in Fig. 4a–c, the asperities of the MoS<sub>2</sub>/DLC-4L coating were flattened in the sliding direction by plastic deformation, revealing a smooth and burnished appearance. The elliptical wear scar on the counterpart steel ball were covered by continuous and compact transfer layer formed from the MoS<sub>2</sub>/DLC-4L coating (Fig. 4d–f). Furthermore, the thickness of transfer layer increases with the increasing applied load, which can explain why the wear rates of the MoS<sub>2</sub>/DLC-4L coating under different applied loads are in the same order of magnitude as shown in Fig. S2 in ESI.† It can be concluded that the applied load has a slightly effect on the tribological behavior of the MoS<sub>2</sub>/DLC-4L coating under vacuum condition without SLD.

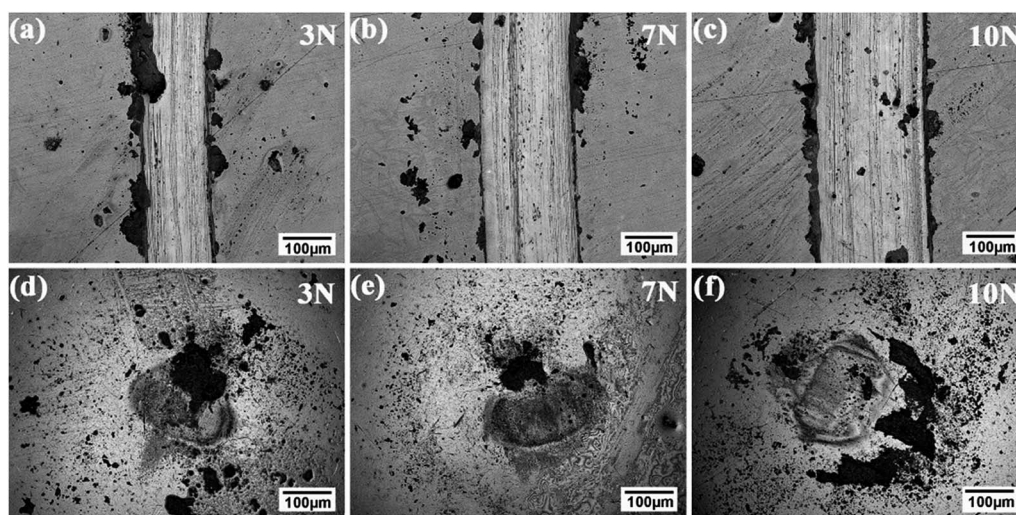


Fig. 4 SEM images of wear tracks for the MoS<sub>2</sub>/DLC-4L coating and wear scars of corresponding counterpart steel balls under different applied loads in a high vacuum environment without SLD. (a, d) 3 N; (b, e) 7 N; (c, f) 10 N.

Typical SEM photographs of the wear tracks for the MoS<sub>2</sub>/DLC-4L coating under different applied loads in a high vacuum environment with SLD are shown in Fig. 5. Compared with the condition without SLD (Fig. 4), the addition of SLD abrasives resulted in the increase in wear to a large extent. The wear tracks are deep and have sharp edges that correlate well with the angular shape of SLD particles. This suggests that the grooves are formed by SLD abrasive particles and not by asperities on the surface of the counterpart steel ball. As shown in Fig. 5a–c, the wear track under light load of 3 N was wider than that under intermediate load (5 N and 7 N), and the edge of wear track did not peel off evenly. It is important to note that most of the SLD particles do not enter into the contact area under light load but flow round the sides of wear track, and only a few particles are seen in the centre of contact area (Fig. 5e). This phenomenon has been reported previously.<sup>28,29</sup> For the applied loads of 5 N and 7 N, the delamination and plastic deformation of the wear track are observed obviously in Fig. 5b and c. In addition, it is observed from the high magnified images in Fig. 5f and g that a large number of small SLD abrasive particles exist in the contact surface, and these micro-/nanoparticles can roll freely to reduce friction (Fig. 3b). However, when the load is up to the maximum value of 10 N (Fig. 5d and h), the load per SLD particle will increase, and thus the stress concentration in the contact region leads to the coating failure due to the severe fatigue wear occurred locally. In terms of the Rabinowicz's derivation of eqn (2),<sup>30</sup> the increase in load has caused the particles to embed more deeply and cutting may become more

significant, which lead to a large increase in wear rate.<sup>31</sup> This coincides with a change in wear mechanism from a three-body rolling process at the light load to a two-body cutting-type process at the heavy load.

$$V = kSN \quad (2)$$

where  $k$  is the wear coefficient with units  $\text{m}^3 (\text{N m})^{-1}$ ,  $S$  can be related to the total distance of sliding,  $N$  is the normal load on the contact,  $V$  is the wear volume.

The 3D morphologies of the wear track for the MoS<sub>2</sub>/DLC-4L coating under the applied load of 7 N in a high vacuum environment with and without SLD are shown in Fig. 6. The 3D morphologies provided the convincing evidences of distinct wear depth and surface characteristics under two lubricated conditions. The wear track was very smooth and shallow under the condition without SLD (Fig. 6a) whereas rough and deep under the condition of SLD (Fig. 6b).

### 3.2. Tribological behaviors of the monolayer and multilayer coatings with SLD

From the above results, the MoS<sub>2</sub>/DLC-4L coating presents excellent tribological performance under intermediate load with SLD. Fig. 7 shows the comparison of tribological properties of pure MoS<sub>2</sub>, DLC and the MoS<sub>2</sub>/DLC-4L coatings under the applied load of 7 N in a high vacuum environment with SLD. In previous reports,<sup>1,32</sup> pure MoS<sub>2</sub> coating under vacuum condition without SLD exhibits excellent tribological performance,

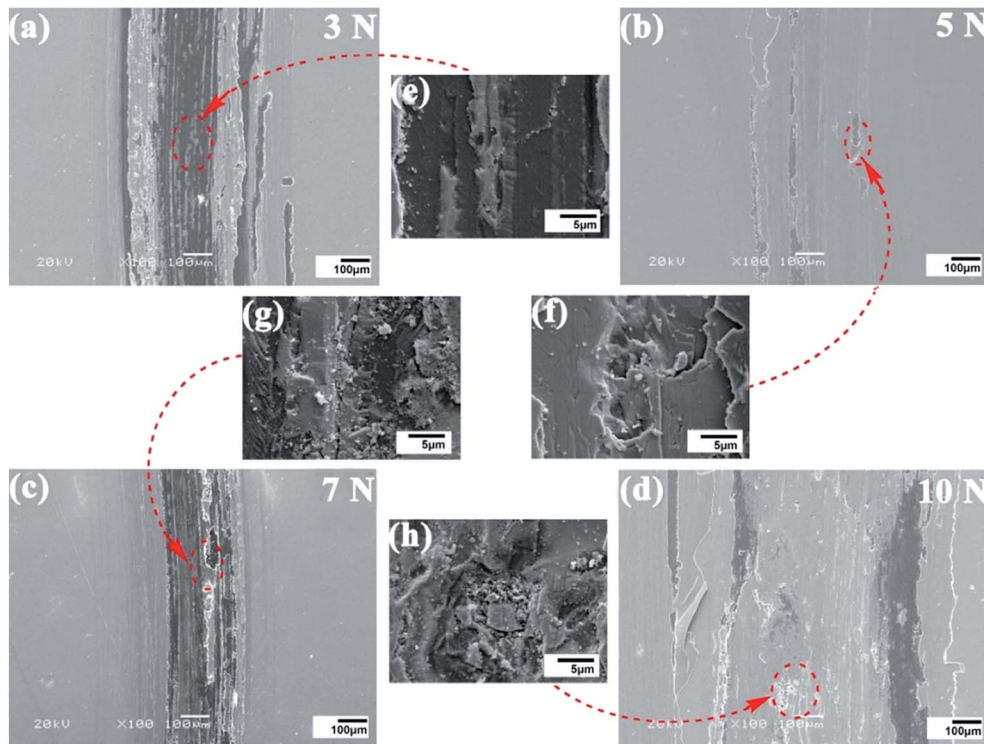


Fig. 5 SEM images of wear tracks under different applied loads in a high vacuum environment with SLD. (a–d) 3 N, 5 N, 7 N, 10 N; (e–h) partial enlarged detail of wear tracks.



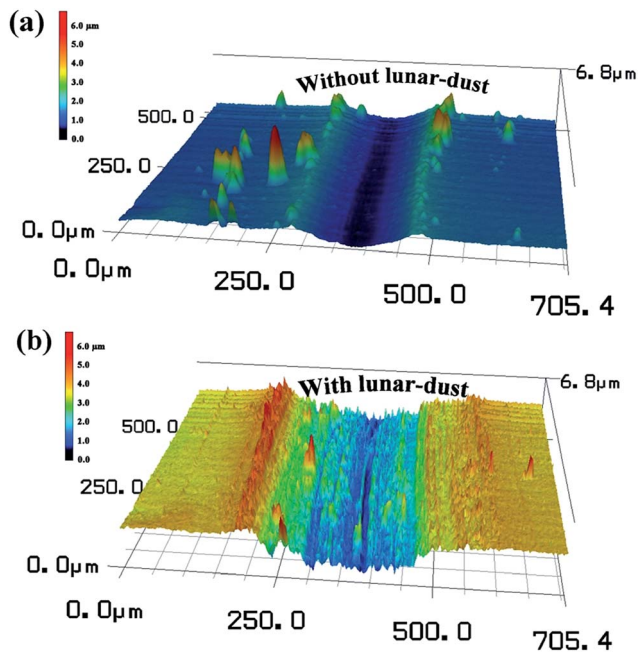


Fig. 6 3D images of wear track under the applied load of 7 N in a high vacuum environment (a) without and (b) with SLD.

whereas the coating only maintains a short lifetime (about 500 cycles) when the SLD are introduced in the contact area. The main reason is that the soft coating is easy to shear off from the substrate due to the mechanical damage by the angular SLD particles. The poor friction behavior of DLC coating can be attributed to the high adhesive effect in vacuum. In contrast, the  $\text{MoS}_2/\text{DLC}$ -4L coating maintains a steady friction coefficient of 0.02 and a longer wear life under high vacuum condition with SLD.

### 3.3. Mechanism of anti-wear and friction-reducing

In order to further understand the tribological behavior and wear mechanism of the  $\text{MoS}_2/\text{DLC}$  multilayer coating with SLD, the effect of the sliding distance on the wear volume for the

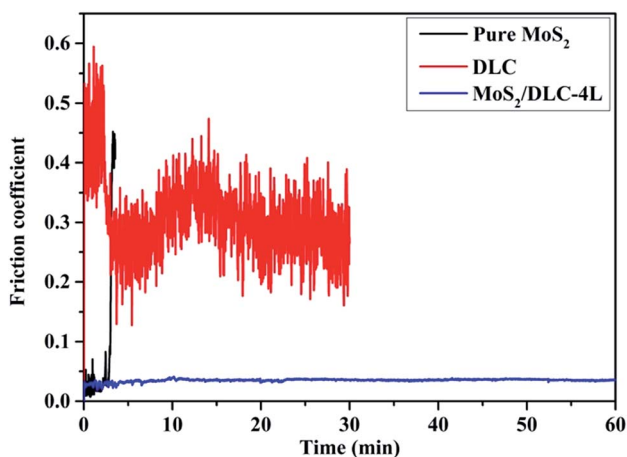


Fig. 7 Friction curves of pure  $\text{MoS}_2$ , DLC and  $\text{MoS}_2/\text{DLC}$ -4L coatings under the applied load of 7 N in a high vacuum environment with SLD.

$\text{MoS}_2/\text{DLC}$ -4L coating against steel ball was examined. The test time was set for 1, 3, 5, 10, 30, and 60 min corresponding to the 180, 540, 900, 1800, 5400, and 10 800 sliding cycles, respectively. The variation in wear rate of the  $\text{MoS}_2/\text{DLC}$ -4L coating is plotted in Fig. 8, as a function of sliding time under the applied load of 7 N in high vacuum conditions (a) without and (b) with SLD. As shown in Fig. 8, in both cases, the initial material loss was significantly higher than the steady-state wear, and the wear rates decreased with the increase of sliding cycle by a power law function. This is a common observation.<sup>5</sup> But the wear mechanism for two cases is totally different. For the condition without SLD, because the surface asperity of counterpart steel ball scratched the coating surface at running-in stage, the transfer layer was not yet form, and resulted in the high wear ratio at initial stage. With increasing sliding cycle, the wear ratio decreased. This was because the counterpart steel ball was polished, and the coating material transferred to contact surface to form gradually a transfer layer. When a compact, dense and continuous tribofilm formed on the counterpart surface, the friction reached the stable wear stage. As showed in Fig. 8a, the wear rates of the  $\text{MoS}_2/\text{DLC}$ -4L coating decreased from  $6.9 \times 10^{-5} \text{ mm}^3 \text{ N}^{-1} \text{ m}^{-1}$  at 180 cycles to  $1.6 \times 10^{-6} \text{ mm}^3$

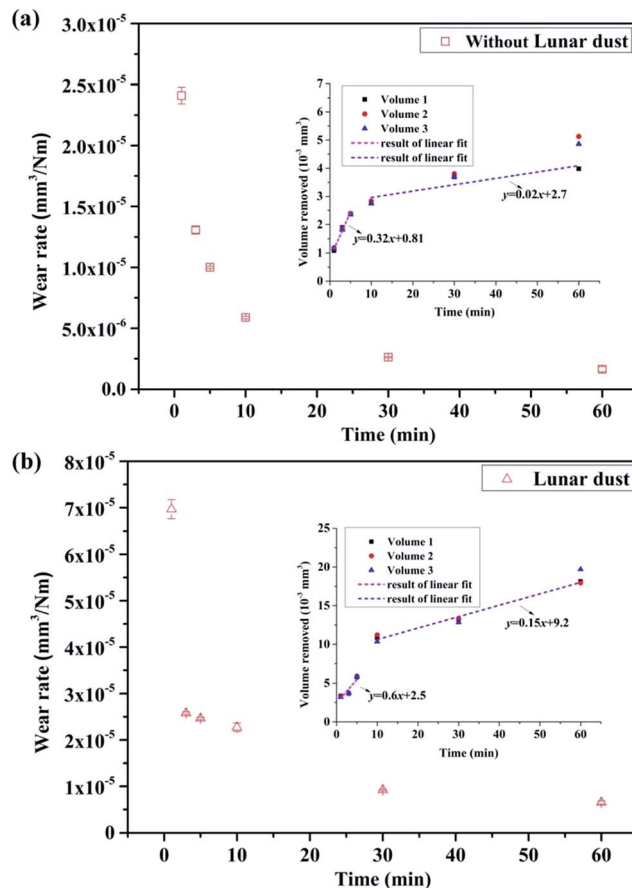


Fig. 8 The wear rates (main drawing) and wear volume (inset) as a function of sliding cycle for the  $\text{MoS}_2/\text{DLC}$ -4L coating under the applied load of 7 N in high vacuum conditions (a) without and (b) with SLD. The inset is fitted the wear volume as a linear function of sliding cycle.

$\text{N}^{-1} \text{m}^{-1}$  at 10 800 cycles. In comparison, the wear ratio with SLD is higher slightly than that without, which indicated the SLD dominated the wear. At the initial sliding stage, the angular SLD particles with high hardness penetrated into the coating surface with soft  $\text{MoS}_2$  top-layer, and scratched the coating surface to form microploughing, microcutting and pitting morphologies. This resulted in high wear ratio. With increasing sliding cycle, the sliding performed on exposed DLC sublayer with high hardness and compact structure. The SLD particles were milled to form small size of micro-/nanoparticles with smooth surface. Amounts of milled SLD wrapped by  $\text{MoS}_2$  could roll freely in groove of the contact region to decrease the wear ratio. As showed in Fig. 8b, the wear rates of the  $\text{MoS}_2/\text{DLC-4L}$  coating decreased from  $7.0 \times 10^{-5} \text{ mm}^3 \text{ N}^{-1} \text{ m}^{-1}$  at 180 cycles to  $6.5 \times 10^{-6} \text{ mm}^3 \text{ N}^{-1} \text{ m}^{-1}$  at 10 800 cycles.

Based on the results of wear rate, we fitted the wear volume as a linear function. The relationship between wear volume and friction time can be illustrated as the inset of Fig. 8a and b. It was found that the total wear volume was directly proportional to the sliding cycle, which could be expressed using the Archard's law mostly used in dry friction and abrasive wear conditions,<sup>29,33</sup> i.e.

$$Q = K \frac{LN}{H}$$

where  $Q$  is volume worn loss,  $L$  is total sliding distance,  $N$  is the applied load,  $K$  is the wear coefficient and  $H$  is the hardness of the softer of the materials in contact. The wear volume increases linearly with sliding time. Moreover, the slopes of wear volume/time at initial stage are higher than the stable stage regardless of frictional conditions. For the dry sliding tests, the high slope at running-in process is probably due to mechanical interlocking generated by the surface asperity of two mating bodies. When the SLD is introduced in the contact, the higher slope at initial stage is mainly attributed to the mechanical scraping from the sharp particles. With the increase of sliding cycle, the particles can be broken and milled into the fine particles to roll, thus resulting in the stable friction coefficient and lower slope than initial stage. This confirms that a transition between the two mechanisms (two-body cutting to three-body rolling) will occurs at the region from initial to stable stage. More interestingly, the slopes for SLD condition are higher than the dry condition under the same stage. We claim that the high value of slope is caused by the abrasive effect of three-body particles.

Fig. 9 shows the photomicrographs of the wear tracks of the  $\text{MoS}_2/\text{DLC-4L}$  coating after 180, 900, 1800, and 10 800 sliding cycles against steel balls under the applied load of 7 N in high vacuum environment with SLD. To get a better look at the abrasive coating surface, the tested samples were cleaned ultrasonically in ethanol bath to remove the cohesive particles. As shown in Fig. 9a, the wear surface was relatively shallow after 180 cycles, indicating that  $\text{MoS}_2$  top layer was still present. Moreover, the wear track showed several microgrooves due to the cutting effect of the angular SLD particles. However, the partial region of the wear track revealed a rougher surface after 900 cycles (Fig. 9b), which was attributed to the delamination of  $\text{MoS}_2$  top layer. When sliding cycle increased to the 1800 cycles

(Fig. 9c), wear process mainly occurred on the DLC sublayer. The central region of the wear track was dominated by indentation/plastic deformation characteristics, which perhaps suggested that the wear mechanisms of DLC layer was a mixed mechanism involving grooving and rolling abrasive wear during the sliding cycle. When the sliding cycle was up to the 10 800 cycles, the wear track revealed severe plastic deformation and some "ridge" layer. In particular, such ridge formation at the centre of the wear track was richer than that at the edge region, which was consistent with the results of the 3D profile of wear track (Fig. 6b). This ridge layer acting as third body reservoirs can achieve long life and low friction to protect worn spots in sliding contact. Meanwhile, the SLD particles could roll freely between the ridge layer and the steel ball surface to reduce friction and wear. Therefore, the wear rate decreases by a power-law function with the sliding cycle (Fig. 8), and the three-body rolling wear is the dominating wear mechanism under the applied load of 7 N, suggesting a lower slope of wear volume/time line at initial stage than stable stage.

The energy dispersive X-ray (EDX) elemental distribution maps of wear track for the  $\text{MoS}_2/\text{DLC-4L}$  coating after 180, 1800, and 10 800 sliding cycles against steel balls under the applied load of 7 N in high vacuum environment with SLD are shown in Fig. 10. The brighter area means a higher concentration of the detected ion. Results indicate that the wear track is rich in some SLD elements (such as O, Si, and Fe) and coating elements, regardless of the sliding cycle. As indicated by the Mo, S, C, and Cr mapping images in Fig. 10, the concentration of Mo and S elements decreased gradually while that of C and Cr elements increased with increasing sliding cycle. As shown in Fig. 10a, the  $\text{MoS}_2$  top layer did not peel off at the initial friction stage. When the sliding cycle rose to 10 800 (Fig. 10c), the Mo and S elements were only presented at the central region of wear track. We suggested that the Mo and S elements were derived from the top layer based on the 3D morphologies of wear track (Fig. 6) and the thickness value of the coating (Fig. 1).

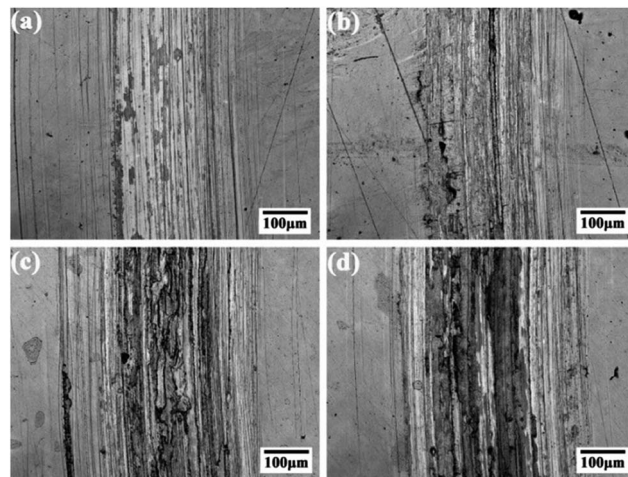


Fig. 9 The photomicrographs of the wear surfaces of the  $\text{MoS}_2/\text{DLC-4L}$  coating under the applied load of 7 N in high vacuum environment with SLD. (a) 180 cycles, (b) 900 cycles, (c) 1800 cycles, (d) 10 800 cycles.



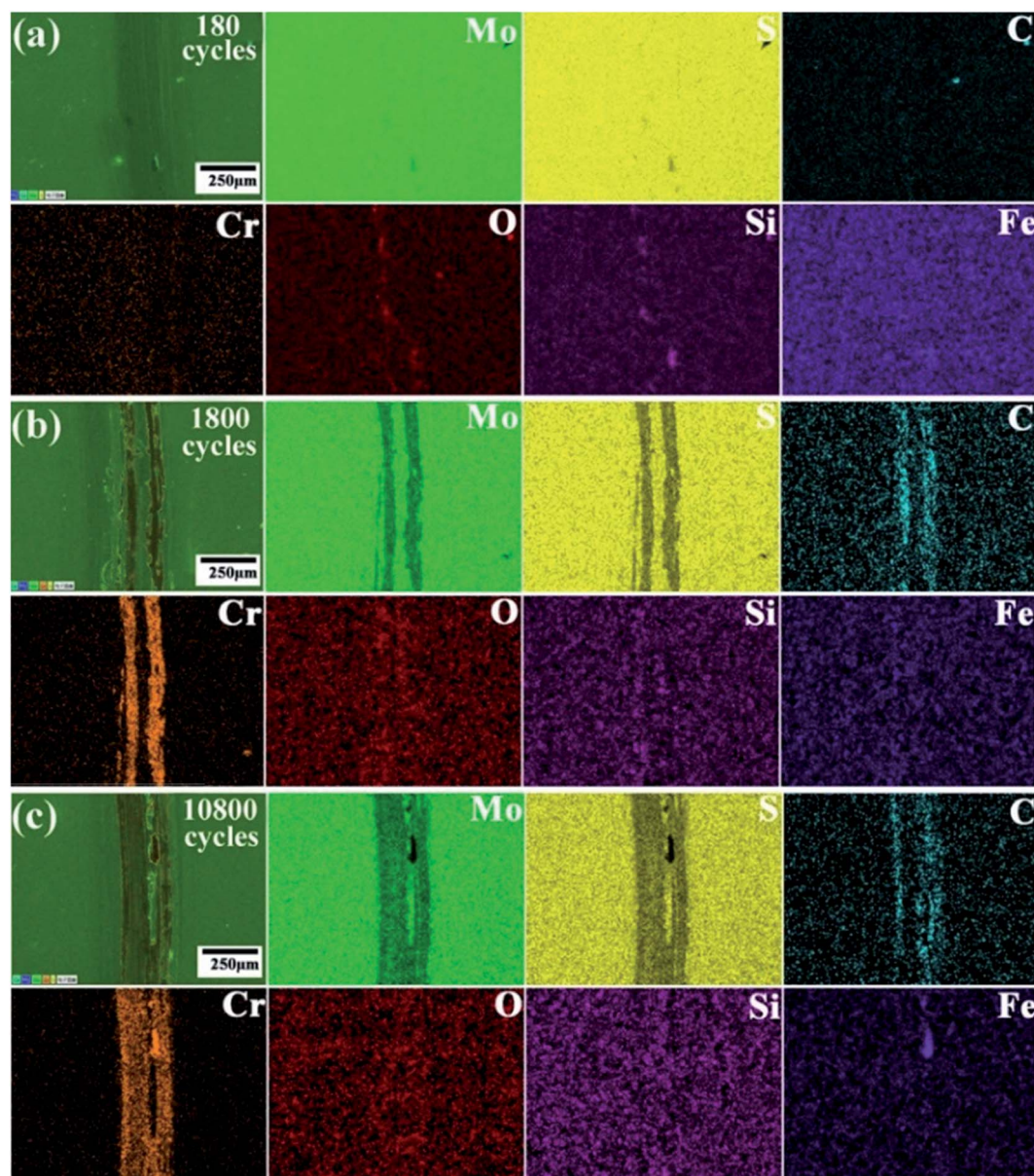


Fig. 10 Elemental distribution maps of wear track of the MoS<sub>2</sub>/DLC-4L coating under the applied load of 7 N in high vacuum environment with SLD. (a) 180 cycles, (b) 1800 cycles, (c) 10 800 cycles.

The Raman spectra provide an insight into the structure of the wear track for the MoS<sub>2</sub>/DLC-4L coating and steel balls. Results showed that two well-defined bands at 383 (E<sub>2g</sub>) and 408 (A<sub>1g</sub>) cm<sup>-1</sup> were assigned to MoS<sub>2</sub> single crystal in all test conditions (Fig. 11).<sup>34,35</sup> In addition, the broadening peak at the approximate 285 and 600 cm<sup>-1</sup> could be related to the presence of MoO<sub>3</sub> because the MoS<sub>2</sub> phase was easily oxidized into the MoO<sub>3</sub> in air after friction test. However, much stronger MoO<sub>3</sub> modes at ~666, 850 and 995 cm<sup>-1</sup> were not observed.<sup>36</sup> The line shapes of Raman spectra acquired from wear tracks and scars after 180, 540, and 900 sliding cycles were similar to each other and to that of the as-deposited coating, *i.e.* it only showed the sharp TMD peaks, and the C peak (1000–1700 cm<sup>-1</sup>) was not observed. When the sliding cycle was up to 1800, the intensity of

the MoS<sub>2</sub> peak decreased whereas that of C peak increased with the increasing sliding cycle (Fig. 11). As seen from the inset in Fig. 11a, when the sliding cycle rose to 5400, *I<sub>D</sub>/I<sub>G</sub>* ratio increased, and the G band peak shifted to higher Raman frequency. This could be attributed to the graphitization process of the coating during the wear process, leading to the phase transition from sp<sup>3</sup> to sp<sup>2</sup>.<sup>13,37</sup> Also, the inset in Fig. 11b showed D-band at 1412 cm<sup>-1</sup> and G-band at 1582 cm<sup>-1</sup> for carbon after 1800 sliding cycles. It was indicated that the friction process occurred in the DLC layer (second layer) and formed a carbon transfer layer on the contact surface of steel ball. Moreover, the intensity ratio between C and MoS<sub>2</sub> increased with the increasing sliding cycle, suggesting that the transfer layer mainly consisted of the disordered graphitic

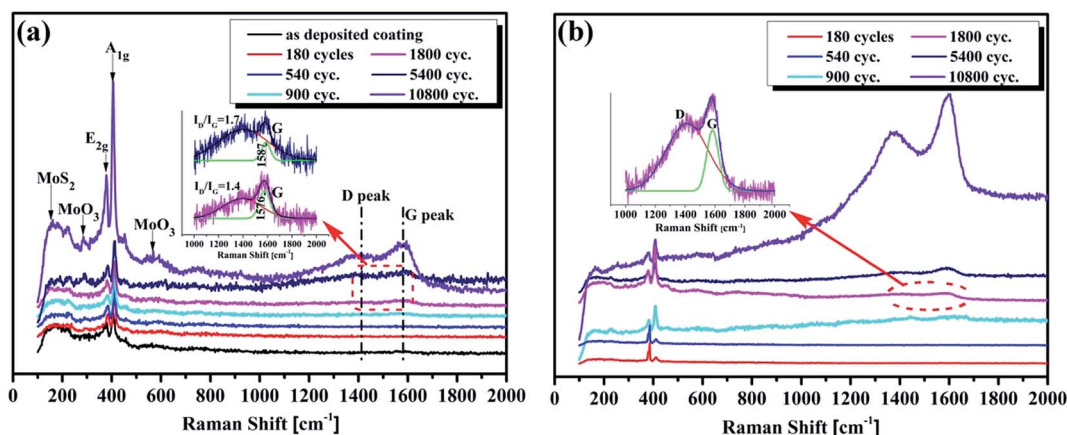


Fig. 11 Raman spectra acquired from different sliding cycles of the wear tracks for the (a) MoS<sub>2</sub>/DLC-4L coating and (b) steel ball under the applied load of 7 N in high vacuum environment with SLD.

carbon and less order MoS<sub>2</sub> after 10 800 cycles. With both species found in the spectrum after sliding, the friction was likely controlled by both species acting synergistically.

The chemical states of the transfer layer tested at different sliding cycles are characterized by XPS spectra. As shown in Fig. 12a, the results indicated the existence of molybdenum, sulphur, carbon, oxygen, silicon, iron, and calcium. It is clear that molybdenum, sulphur and carbon are from the coating material. Oxygen, silicon, and calcium derive from the SLD. As for oxygen element, the as-deposited MoS<sub>2</sub>/DLC-4L coating is mainly derived from the absorbed oxygen of coating. However, when the coating is tested on the SLD environment, the position of O 1s is shifted to lower binding energy. This indicates that the oxide of transfer layer is mainly from the SLD particles and the effect of mating material. As for iron element, a large amount of iron comes from the steel ball while small amount comes from the particles. Moreover, except for the Mo and S peaks, the peak shapes of other elements are very similar regardless of the sliding cycle, indicating that the SLD has small chemical changes. In order to further analyse the chemical states, the C 1s, S 2p and Mo 3d spectra of the coating as well as the O 1s, Si 2p, Fe 2p, and Ca 2p spectra of the SLD are fitted by Gaussian-Lorentzian function. The C 1s spectra can be fitted into five distinct carbon species: C-C species at 285.4 eV, C=C species at 284.6 eV, C-O-C species at 286.4 eV, and C=O species at 288.5 eV. According to literature reports, in case of the friction conditions with SLD, the C 1s peaks show contribution from carbon bonded to silicon (C-Si) at 284.1 eV, which is in agreement with the Si 2p spectra.<sup>38</sup> The broad peaks of Si 2p around 102.1 and 102.8 eV are correspond to SiC and SiO<sub>x</sub>, respectively. This suggests that a containing SiC composite transfer layer forms on steel ball from the MoS<sub>2</sub>/DLC-4L coating and SLD. The O 1s peak at 532.0 eV is mainly attributed to the absorbed oxygen on the coating surface or the oxide of the introduced SLD. Fig. 12b and c show the deconvoluted Mo 3d and S 2p XPS spectra. Mo 3d spectra are fitted into four components, representing the MoS<sub>2</sub> (228.9 and 232.2 eV) and MoO<sub>x</sub> (228.1 and 231.6 eV), respectively. After 180 and 900 sliding cycles, a small

shoulder at approximately 226.0 eV assigning to S 2s appears in Mo 3d spectra.<sup>39-41</sup> Moreover, the fitted results and peak shapes of Mo and S elements are very similar to the as-deposited coating, which indicates that the structure of the transfer layer does not change obviously. The reason is that the transfer layer is mainly composed of MoS<sub>2</sub> from the mechanical abrasion and the SLD particles. With the increase in sliding cycle, the Mo-C bond is also observed at 227.8 and 231.0 eV,<sup>39</sup> due to the increases chemical action of friction interface. It indicates that the MoS<sub>2</sub> transferring to the steel ball can passivate the friction-induced carbon dangling bonds to some extent, which can prevent the adhesion and cold welding between the DLC layer and the counterpart under high vacuum. Thus, the solid compact wear debris layer on the top of wear scars after sliding 10 800 cycles is mainly dominated by MoS<sub>2</sub>/Mo-C structure under the repeated action of compression.

Based on the fitting results of XPS spectra, the friction behavior and the wear mechanism of the selected multilayer coating under vacuum condition with SLD can be further explained. In the initial sliding stage of 180 and 900 cycles, the SLD particles could become trapped between the sliding surfaces and scratched the coating surface to form micro-ploughing, microcutting and pitting morphologies due to the relative low hardness of top layer (Fig. 9), thereby increasing the wear rate (Fig. 8). At the same time, the SLD particles started to penetrate into the contact surface of steel ball and mixed with some coating debris to form a composite transfer layer at initial friction stage. With the increasing sliding cycle, the formed hard molybdenum carbide phase in the composite transfer layer found by XPS which could improve the anti-wear performance of the coating due to decreasing contact area to shear. Furthermore, the SLD was polished to the flat facets and formed impact nanoparticles stack at the stable stage, which further decreased the contact stress between SLD and coating and the wear rate of coating.

Fig. 13 shows the major chemical composition of the composite transfer layer as a function of the sliding cycle. Results showed that the Mo and S elements of coating



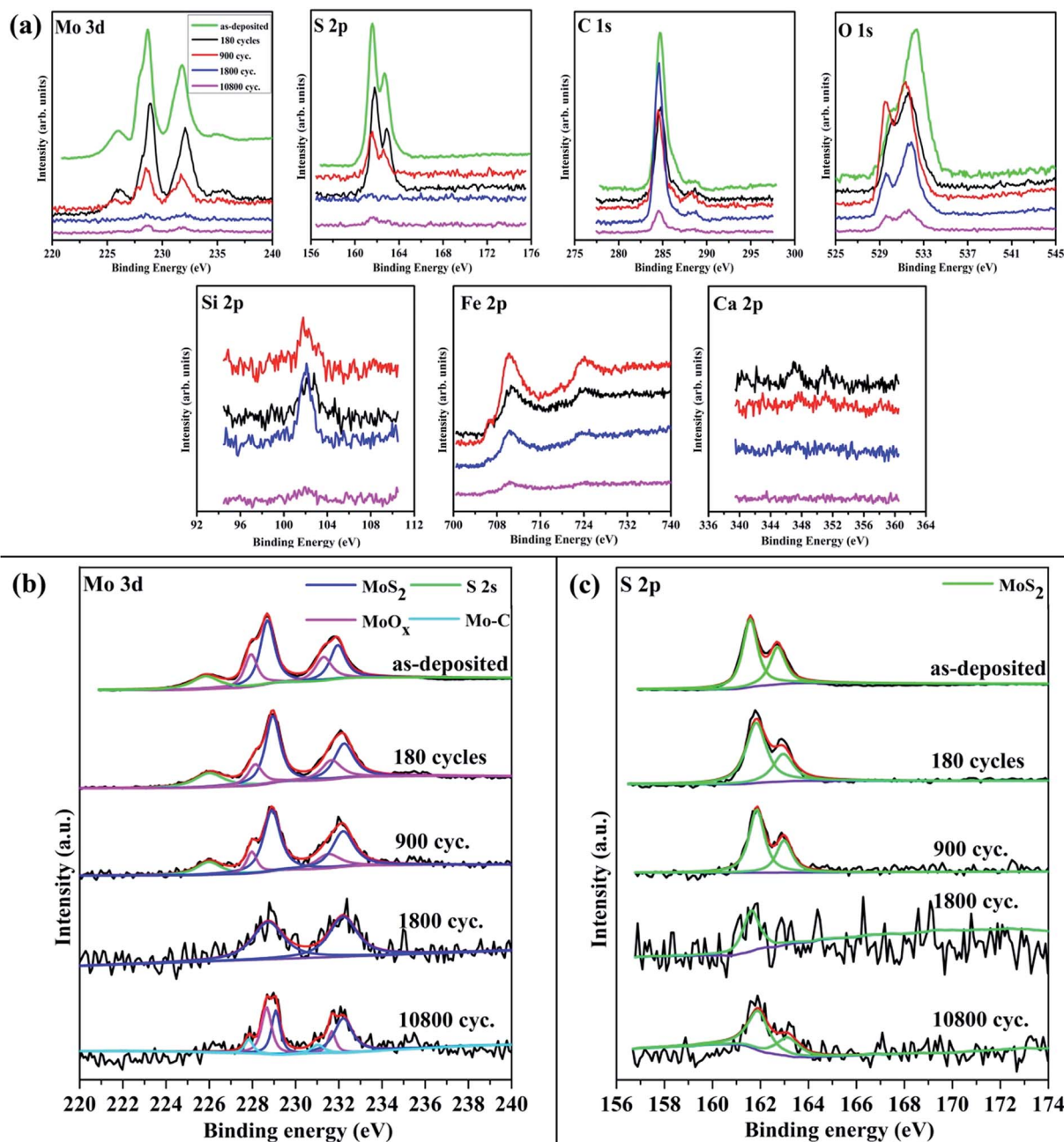


Fig. 12 (a) XPS spectra of Mo 3d, S 2p, C 1s, O 1s, Si 2p, Fe 2p, and Ca 2p obtained from the transfer layer on the steel balls for different sliding cycles under the applied load of 7 N in high vacuum environment with SLD (the green line obtained from the as-deposited MoS<sub>2</sub>/DLC-4L coating); (b) deconvoluted Mo 3d XPS spectra; (c) deconvoluted S 2p XPS spectra.

decreased first and increased afterward with the increase in sliding cycle. However, the carbon element from the hard layer (DLC) had an opposite tendency that the composition increased first and decreased afterward with the increasing cycle. Interestingly, the Si, Al elements from the SLD enriched on the contact surface and their contents increased monotonically with the increasing sliding cycle. Therefore, it is demonstrated that the formed transfer layer is a composite transfer layer consisted of SLD particles and wear debris of the MoS<sub>2</sub>/DLC-4L coating. This composite transfer layer acts as a protective layer to depress the wear of the steel ball and coating.<sup>42</sup> Inset in

Fig. 13 shows the evolution of the S/Mo ratio as a function of the sliding cycle. The S/Mo ratio increased dramatically in the initial stage and then increased gradually with the increasing sliding cycle. It is anticipated that friction-induced re-orientation of basal planes occurs at the beginning of sliding. The relative low S/Mo ratio of as-deposited coating is due to the S can be easier to sputter out of the surface than Mo during deposition of the coating. With the increase in sliding cycle, the S/Mo ratio of the transfer layer increases and is much closer to stoichiometric MoS<sub>2</sub>. Considering the theory proposed by



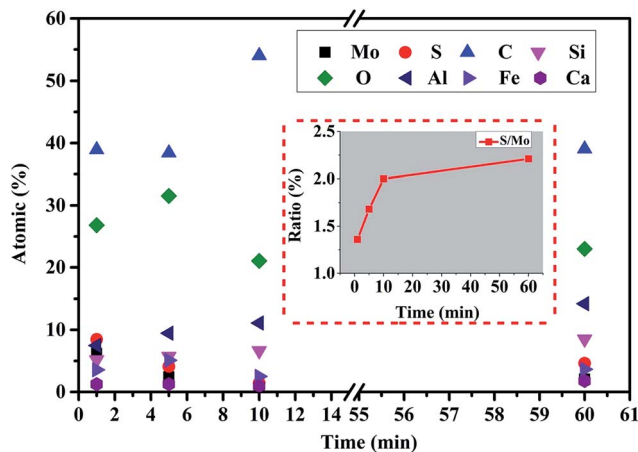


Fig. 13 The variation of chemical composition (main drawing) and S/Mo ratio (inset) of the composite transfer layer as a function of the sliding cycle for the MoS<sub>2</sub>/DLC-4L coating under the applied load of 7 N in high vacuum environment with SLD.

Hirano and Martin, the stoichiometric MoS<sub>2</sub> is more advantageous to reduce friction coefficient in vacuum.<sup>43,44</sup>

The wear mechanism to describe the wear process as the generation and removal of wear debris from the contact surfaces was proposed.<sup>36</sup> Fig. 14a shows the transmission electron microscopy (TEM) image of the wear debris from the test conducted at 7 N for the MoS<sub>2</sub>/DLC-4L coating under high vacuum condition with SLD after 10 800 cycles. According to the corresponding EDX (Fig. 14b), the debris is mainly made of the elements in SLD and MoS<sub>2</sub>, which is consistent with that of the

formed composite transfer layer (confirmed by XPS in Fig. 12). The origin of carbon detected is not well defined: it can either come from TEM grids or from the coating materials. Friction-induced reorientation of the MoS<sub>2</sub> basal planes in the contact interface is clearly demonstrated by high resolution TEM studies. As shown in Fig. 14c, the periphery of the wear debris aggregates some MoS<sub>2</sub> crystals whose most visible feature is the stacking of the (0002) planes with  $d_{0002} = 6.5 \text{ \AA}$ , which can be confirmed by Fig. 14d. The superimposed MoS<sub>2</sub> crystals with a rotation angle are liberated in the contact area, and finally an efficient orientation of these crystals into the ideal angle with the easy shear planes parallels to the sliding direction, as already reported in the literature.<sup>44–46</sup> Moreover, the central region of wear debris presents a polycrystalline structure (Fig. 14e) with the corresponding diffraction pattern of the SLD particles (Fig. 14f). Therefore, the SLD particles are broken under contact pressure and wrapped by MoS<sub>2</sub> with lamellar structure because of the shear stress. An efficient composite layer containing hybrid nanocrystals (SLD nanoparticles and MoS<sub>2</sub>) is eventually formed in the contact interface, which favours low friction and long lifetime of the MoS<sub>2</sub>/DLC-4L coating. This result brings a lot of possibilities for the use of the MoS<sub>2</sub>/DLC multilayer coating in reality space environment.

## 4. Conclusions

We fabricated a series of MoS<sub>2</sub>/DLC multilayer coatings with different bilayer periods by unbalanced magnetron sputtering. The MoS<sub>2</sub>/DLC-4L coating had the optimum comprehensive performance. We also investigated the friction and wear

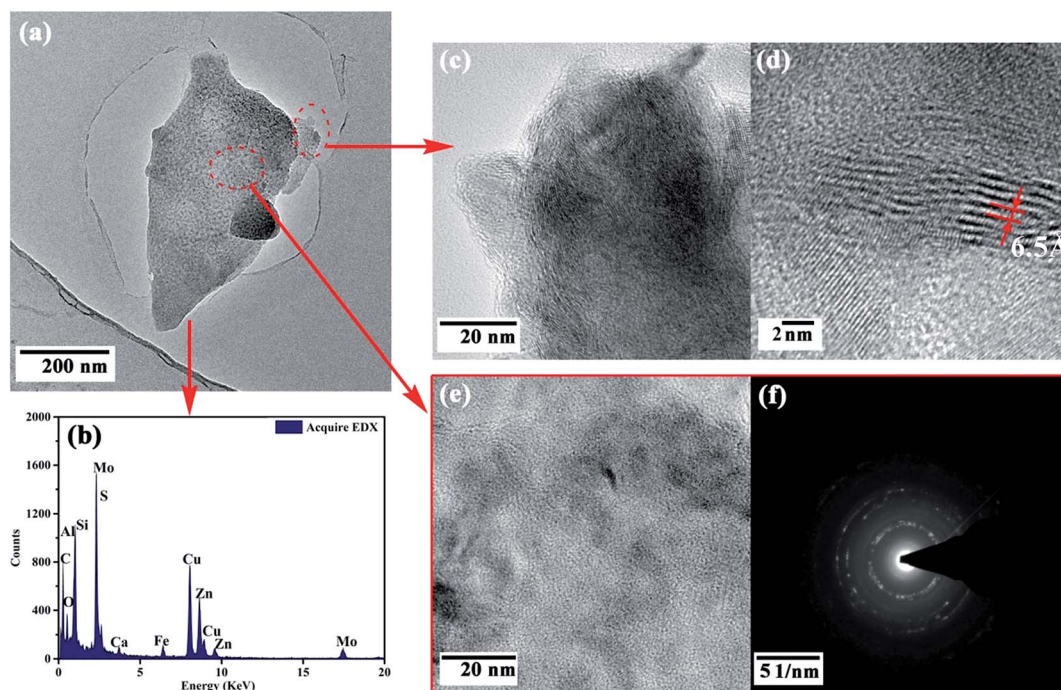


Fig. 14 (a) TEM photograph and (b) corresponding EDX analysis of the wear debris from the test conducted at 7 N for the MoS<sub>2</sub>/DLC-4L coating under high vacuum condition with SLD after 10 800 cycles, (c–e) HRTEM images and (f) SAED pattern of the marked area in (a).

behaviours of MoS<sub>2</sub>/DLC-4L coating under high vacuum condition with and without SLD. It is noteworthy that the wear volume of MoS<sub>2</sub>/DLC-4L coating with the increase of sliding time follows Archard laws, regardless of the condition with or without SLD. Moreover, the MoS<sub>2</sub>/DLC-4L coating in SLC condition exhibited lower friction coefficient and wear rate than those of pure MoS<sub>2</sub> or DLC coating. Characterization of wear tracks by SEM and 3D morphologies revealed that a low wear rate was attributed to the formation of ridge layer at the central region of wear track, where the milled SLD particles could roll freely between the ridge layer and the steel ball surface. Results of wear tracks/scars by XPS and wear debris by TEM showed that the formation of tribo-induced composite transfer layer in the contact interface and the reorientation of the MoS<sub>2</sub> basal planes could effectively reduce the friction and wear.

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